



**MINISTRY OF TRANSPORT OF THE RUSSIAN FEDERATION
FEDERAL AIR TRANSPORT AGENCY**

Type Certificate Data Sheet

№ FATA-AS355

Models:

- AS355E
- AS355F
- AS355F1
- AS355F2
- AS355N
- AS355NP

**Issue 02
30 of March 2018**

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This Data Sheet which is the integral part of Type Certificate № 112-3505. It prescribes the conditions and limitations under which the product for which the type certificate was issued meets the requirements of Certification Basis.

1. Helicopter model AS 355E

Type Certificate Holder	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engines and skid landing gear
Category	Normal
Applicability	AS 355E helicopter model is approved for VFR and IFR day and night operation, above land and water surface, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certification Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type Design	<p>Defined in the following documents:</p> <ul style="list-style-type: none"> – Flight Manual – AS355E, AS355F, AS355F1, AS355F2, AS355N; – Service Manual – AS355; – Overhaul Manual – 355; – Repair Manual – AS355; – AS355 Service Bulletins approved by EASA. <p>The helicopter must be equipped with the following equipment (approved list of mandatory equipment is included in EUROCOPTER FRANCE document №350A04.4320 based on the following list):</p> <ul style="list-style-type: none"> – Pressure altimeter (in meters); – Vertical speed indicator (in m/s); – Attitude Indicator (horizon) with glide indication; – Aircraft clock; – Automatic Direction Finder (ADF). <p>The helicopter should be equipped with emergency VHF radio P855A1</p> <p>Markings for emergency equipment must be in Russian language.</p>

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Certification Basis

CB355.27
 Certification Basis includes requirements to
 Airworthiness AP-27, requirements to Environment
 AP-36

Engine

Two gas-turbine engines
 Allison 250-C20F
 manufactured by Allison Engine Company, Inc.
 Engine Type Certificate
 №83-Д/01 dated on 11 November 1997 Issued by IAC
 AR

Fuel

PT, TC-1 in accordance with GOST 10227-86
 (foreign fuels types are listed in RFM)
 Anti-ice additives: fluid "И" (GOST 8313), "И-М" (TU 6-
 10-1458), volume concentration 0.10-0.30.

Approved oil types for engine and transmission gearbox

see in RFM

Allison 250-C20F Engine operational limits

Modes/ Characteristic	Takeoff mode	Maximum continuous power	
		AEO	OEI
Torque, N×m (%)	406 (78)	380 (73)	521 (100)
Turbine outlet temperature (°C)	810	738	810
Gas generator speed (%)	105	105	105

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №83-Д/01 dated on 11 November 1997 issued by IAC AR on engines Allison 250-C20F Allison Engine Company, Inc.

Rotor Limitations

Power on:

- AEO 390 (+4/-5) rpm
- OEI 375 - 394 rpm

Power off:

- maximum 425 rpm
- minimum 330 rpm (aural warning at 330 rpm)

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Speed limitations

Power on:

Vne is limited by indicated air speed:

150 kt (287 km/h) from 0 m altitude.

With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).

When OAT is lower than - 35°C Vne is additionally decreased by 19 km/h (10 knots).

Power off:

Vne is limited by indicated air speed:

120 kt (222 km/h) from 0 m altitude.

With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).

When OAT is lower than - 25°C Vne is additionally decreased by 37 km/h (20 knots) except when Vne is lower than 120 km/h (65 knots).

C.G. Reference

Longitudinal: - 3.4 m forward of the MRH centerline

Lateral: Aircraft symmetry plane

Minimum crew

1 pilot in R.H. seat

Maximum take-off weight

2100 kg

Fuel capacity

736,7 liters

Number of seats

5

6 - if the aircraft is fitted with the forward dual passenger seat, approved layout is contained in document №355A04.3122

Maximum baggage weight

In RH. side hold 100 kg

In LH. side hold 120 kg

In rear hold 80 kg

On cabin floor:

Forward section 150 kg

Rear section 310 kg

Maximum operational altitude

4875 m (16 000 feet)

OAT temperature range

-40°C ...MCA+35°C (Max +50°C).

Additional operational conditions limitations and information of AS355 E helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

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2. Helicopter model AS 355F

Type Certificate Holder	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engine and skid landing gear
Category	Normal
Applicability	AS 355F helicopter model is approved for VFR and IFR day and night operation, above land and water surface, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certification Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type Design	<p>Defined in the following documents:</p> <ul style="list-style-type: none"> – Flight Manual – AS355E, AS355F, AS355F1, AS355F2, AS355N; – Service Manual – AS355; – Overhaul Manual – 355; – Repair Manual – AS355; – AS355 Service Bulletins approved by EASA. <p>The helicopter must be equipped with the following equipment (approved list of mandatory equipment is included in EUROCOPTER FRANCE document №350A04.4320 based on the following list):</p> <ul style="list-style-type: none"> – Pressure altimeter (in meters); – Vertical speed indicator (in m/s); – Attitude Indicator (horizon) with glide indication; – Aircraft clock; – Automatic Direction Finder (ADF). <p>The helicopter should be equipped with emergency VHF radio P855A1</p> <p>Markings for emergency equipment must be in Russian language.</p>
Certification Basis	CE355.27 Certification Basis includes requirements to Airworthiness AP-27, requirements to Environment AP-36

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Engine

Two gas-turbine engines
Allison 250-C20F
manufactured by Allison Engine Company, Inc.
Engine Type Certificate
№83-Д/01 dated on 11 November 1997 Issued by IAC
AR

Fuel

PT, TC-1 in accordance with GOST 10227-86
(foreign fuels types are listed in RFM)
Anti-ice additives: fluid "И" (GOST 8313), "И-М" (TU 6-10-1458), volume concentration 0.10-0.30%.

Approved oil types for engine and transmission gearbox

see in RFM

Allison 250-C20F Engine operational limits

Modes/ Characteristic	Takeoff mode	Maximum continuous power	
		AEO	OEI
Torque, N×m (%)	406 (78)	380 (73)	521 (100)
Turbine outlet temperature (°C)	810	738	810
Gas generator speed (%)	105	105	105

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №83-Д/01 dated on 11 November 1997 issued by IAC AR on engines Allison 250-C20F Allison Engine Company, Inc.

Rotor Limitations

Power on:

- AEO 390 (+4/-5) rpm
- OEI 375 - 394 rpm

Power off:

- maximum 425 rpm
- minimum 330 rpm (aural warning at 330 rpm)

Speed limitations

Power on:

Vne is limited by indicated air speed:
150 kt (287 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 35°C Vne is additionally decreased by 19 km/h (10 knots).

Power off:

Vne is limited by indicated air speed:
120 kt (222 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 25°C Vne is additionally decreased by 37 km/h (20 knots) except when Vne is lower than 120 km/h (65 knots).

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C.G. Reference	Longitudinal: - 3.4 m forward of the MRH centerline Lateral: Aircraft symmetry plane
Minimum crew	1 pilot in R.H. seat
Maximum take-off weight	2300 kg
Fuel capacity	736,7 liters
Number of seats	5 6 - if the aircraft is fitted with the forward dual passenger seat, approved layout is contained in document №355A04.3122
Maximum baggage weight	In R.H. side hold 100 kg In L.H. side hold 120 kg In rear hold 80 kg On cabin floor: Forward section 150 kg Rear section 310 kg
Maximum operational altitude	4875 m (16 000 feet)
OAT temperature range	-40°C ...MCA+35°C (Max +50°C)

Additional operational conditions limitations and information of AS355 F helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

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3. Helicopter model AS 355F1

Type Certificate Holder:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engines and skid landing gear
Category	Normal
Applicability	AS 355F1 helicopter model is approved for VFR and IFR day and night operation, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certification Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type Design	<p>Defined in the following documents:</p> <ul style="list-style-type: none"> – Flight Manual – AS355E, AS355F, AS355F1, AS355F2, AS355N; – Service Manual – AS355; – Overhaul Manual – 355; – Repair Manual – AS355; – AS355 Service Bulletins approved by EASA. <p>The helicopter must be equipped with the following equipment (approved list of mandatory equipment is included in EUROCOPTER FRANCE document №350A04.4320 based on the following list):</p> <ul style="list-style-type: none"> – Pressure altimeter (in meters); – Vertical speed indicator (in m/s); – Attitude Indicator (horizon) with glide indication; – Aircraft clock; – Automatic Direction Finder (ADF). <p>The helicopter should be equipped with emergency VHF radio P855A1</p> <p>Markings for emergency equipment must be in Russian language.</p>
Certification basis	CE355.27 Certification Basis includes requirements to Airworthiness AP-27, requirements to Environment AP-36

Title	Issue	Date
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Engine

Two gas-turbine engines
Allison 250-C20F
manufactured by Allison Engine Company, Inc.
Engine Type Certificate
№83-Д/01 dated on 11 November 1997 Issued by IAC
AR

Fuel

PT, TC-1 in accordance with GOST 10227-86
(foreign fuels types are listed in RFM)
Anti-ice additives: fluid "И" (GOST 8313), "И-М" (TU 6-10-1458), volume concentration 0.10-0.30%.

Approved oil types for engine and transmission gearbox

see in RFM

Allison 250-C20F Engine operational limits

Modes/ Characteristic	Takeoff mode	Maximum continuous power	
		AEO	OEI
Torque, N×m (%)	406 (78)	380 (73)	521 (100)
Turbine outlet temperature (°C)	810	738	810
Gas generator speed (%)	105	105	105

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №83-Д/01 dated on 11 November 1997 issued by IAC AR on engines Allison 250-C20F Allison Engine Company, Inc.

Rotor Limitations

Power on:

- AEO 390 (+4/-5) rpm
- OEI 375 - 394 rpm

Power off:

- maximum 425 rpm (aural warning at 410 rpm)
- minimum 330 rpm (aural warning at 360 rpm)

Speed limitations

Power on:

Vne is limited by indicated air speed:
150 kt (287 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 35°C Vne is additionally decreased by 19 km/h (10 knots).

Power off:

Vne is limited by indicated air speed:
120 kt (222 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 25°C Vne is additionally decreased by 37 km/h (20 knots) except when Vne is lower than 120 km/h (65 knots).

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C.G. Reference	Longitudinal: - 3.4 m forward of the MRH centerline Lateral: Aircraft symmetry plane
Minimum crew	1 pilot in R.H. seat
Maximum take-off weight	2400 kg
Fuel capacity	736,7 liters
Number of seats	5 6 - if the aircraft is fitted with the forward dual passenger seat, approved layout is contained in document №355A04.3122
Maximum baggage weight	In R.H. side hold 100 kg In L.H. side hold 120 kg In rear hold 80 kg On cabin floor: Forward section 150 kg Rear section 310 kg
Maximum operational altitude	4875 m (16 000 feet)
OAT temperature range	-40 °C...MCA +35 °C (Max +50 °C)

Additional operational conditions limitations and information of AS355 F helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

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4. Helicopter model AS 355F2

Type Certificate Holder:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engines and skid landing gear
Category	Normal
Applicability	AS 355F2 helicopter model is approved for VFR and IFR day and night operation, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certificate Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type design	<p>Defined in the following documents:</p> <ul style="list-style-type: none"> – Flight Manual – AS355E, AS355F, AS355F1, AS355F2, AS355N; – Service Manual – AS355; – Overhaul Manual – 355; – Repair Manual – AS355; – AS355 Service Bulletins approved by EASA. <p>The helicopter must be equipped with the following equipment (approved list of mandatory equipment is included in EUROCOPTER FRANCE document №350A04.4320 based on the following list):</p> <ul style="list-style-type: none"> – Pressure altimeter (in meters); – Vertical speed indicator (in m/s); – Attitude Indicator (horizon) with glide indication; – Aircraft clock; – Automatic Direction Finder (ADF). <p>The helicopter should be equipped with emergency VHF radio P855A1</p> <p>Markings for emergency equipment must be in Russian language.</p>
Certification Basis	CE355.27 Certification Basis includes requirements to Airworthiness AP-27, requirements to Environment AP-36

Title	Issue	Date
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Engine

Two gas-turbine engines
Allison 250-C20F
manufactured by Allison Engine Company, Inc.
Engine Type Certificate
№83-Д/01 dated on 11 November 1997 Issued by
IAC AR

Fuel

PT, TC-1 in accordance with GOST 10227-86
(foreign fuels types are listed in RFM)
Anti-ice additives: fluid "И" (GOST 8313), "И-М" (ТУ
6-10-1458), volume concentration 0.10-0.30%.

Approved oil types for engine and transmission gearbox

See in RFM

Allison 250-C20F Engine operational limits

Modes/ Characteristic	Takeoff mode	Maximum continuous power	
		AEO	OEI
Torque, N×m (%)	406 (78)	380 (73)	521 (100)
Turbine outlet temperature (°C)	810	738	810
Gas generator speed (%)	105	105	105

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №83-Д/01 dated on 11 November 1997 issued by IAC AR on engines Allison 250-C20F Allison Engine Company, Inc.

Rotor Limitations

Power on:

- AEO 390 (+4/-5) rpm
- OEI 375 - 394 rpm

Power off:

- maximum 425 rpm (aural warning at 410 rpm)
- minimum 330 rpm (aural warning at 360 rpm)

Speed limitations

Power on:

Vne is limited by indicated air speed:
150 kt (287 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h
per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 35°C Vne is additionally
decreased by 19 km/h (10 knots).

Power off:

Vne is limited by indicated air speed:
120 kt (222 km/h) from 0 m altitude.
With altitude increasing, it is decreased by 15 km/h
per each 1000 m (2,5 kt per 1000 ft).
When OAT is lower than - 25°C Vne is additionally
decreased by 37 km/h (20 knots) except when Vne is
lower than 120 km/h (65 knots).

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C.G. Reference	Longitudinal: - 3.4 m forward of the MRH centerline Lateral: Aircraft symmetry plane
Minimum crew	1 pilot in R.H. seat
Maximum take-off weight	2540 kg
Fuel capacity	736,7 liters
Number of seats	5 6 - if the aircraft is fitted with the forward dual passenger seat, approved layout is contained in document №355A04.3122
Maximum baggage weight	In R.H. side hold 100 kg In L.H. side hold 120 kg In rear hold 80 kg On cabin floor: Forward section 150 kg Rear section 310 kg
Maximum operational altitude	4875 m (16 000 feet)
OAT temperature range	-40 °C...MCA +35 °C (Max +50 °C)

Additional operational conditions limitations and information of AS355 F helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

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5. Helicopter model AS 355N

Type Certificate Holder:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engines and skid landing gear
Category	Normal
Applicability	AS 355N helicopter model is approved for VFR and IFR day and night operation, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certificate Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type design	<p>Defined in the following documents:</p> <ul style="list-style-type: none"> – Flight Manual – AS355E, AS355F, AS355F1, AS355F2, AS355N; – Service Manual – AS355; – Overhaul Manual – 355; – Repair Manual – AS355; – AS355 Service Bulletins approved by EASA. <p>The helicopter must be equipped with the following equipment (approved list of mandatory equipment is included in EUROCOPTER FRANCE document №350A04.4320 based on the following list):</p> <ul style="list-style-type: none"> – Pressure altimeter (in meters); – Vertical speed indicator (in m/s); – Attitude Indicator (horizon) with glide indication; – Aircraft clock; – Automatic Direction Finder (ADF). <p>The helicopter should be equipped with emergency VHF radio P855A1</p> <p>Markings for emergency equipment must be in Russian language.</p>
Certification basis	CE355.27 Certification Basis includes requirements to Airworthiness AP-27, requirements to Environment AP-36

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Engine

Two gas-turbine engines
 Arrius 1A
 manufactured by Turbomeca
 Engine Type Certificate
 № 111-Д/01 dated on 28 October 1996 Issued by
 IAC AR

Fuel

PT, TC-1 in accordance with GOST 10227-86
 (foreign fuels types are listed in RFM)
 Anti-ice additives: fluid "И" (GOST 8313), "И-М"
 (TU 6-10-1458), volume concentration 0.10-0.30%
 %.
 Anti-static Sigbol is approved for use in amounts up
 to 0.0005% by weight.
 see in RFM

Approved oil types for engine and transmission gearbox

Arrius 1A engine operational limits:

Power mode	Shaft torque limit, N×m (%)	Maximum turbine outlet temperature °C	Maximum gas generator speed rpm
Takeoff power	406 (78)	800	54685
Middle emergency power (30 min)	559 (115)	800	55300
Maximum emergency power (2 min 30 sec)	689 (131)	870	56140
Maximum continuous power (AOE)	380 (73)	765	53285
Maximum continuous power (OEI)	521 (100)	765	53285

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №111-Д/01 dated on 28 October 1996 issued by IAC AR on engines TURBOMECA Arrius 1A.

Rotor Limitations

Power on:

- AOE 390 (+4/-5) rpm
- At speed less than 55 knots 390 (+10/-5) rpm
- OEI 375 - 394 rpm

Power off:

- maximum 425 rpm (aural warning at 410 rpm)
- minimum 330 rpm (aural warning at 360 rpm)

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Speed limitations

Power on:

Vne is limited by indicated air speed:

150 kt (287 km/h) from 0 m altitude.

With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).

When OAT is lower than - 35°C Vne is additionally decreased by 19 km/h (10 knots).

Power off:

Vne is limited by indicated air speed:

120 kt (222 km/h) from 0 m altitude.

With altitude increasing, it is decreased by 15 km/h per each 1000 m (2,5 kt per 1000 ft).

When OAT is lower than - 25°C Vne is additionally decreased by 37 km/h (20 knots) except when Vne is lower than 120 km/h (65 knots).

C.G. Reference

Longitudinal: - 3.4 m forward of the MRH centerline

Lateral: Aircraft symmetry plane

Minimum crew

1 pilot in R.H. seat

Maximum take-off weight

2540 kg

Fuel capacity

736,7 liters

Number of seats

5

6 - if the aircraft is fitted with the forward dual passenger seat, approved layout is contained in document №355A04.3122

Maximum baggage weight

In R.H. side hold 100 kg

In L.H. side hold 120 kg

In rear hold 80 kg

On cabin floor:

Forward section 150 kg

Rear section 310 kg

Maximum operational altitude

4875 m (16 000 feet)

OAT temperature range

-40 °C...MCA +35 °C (Max +50 °C)

Additional operational conditions limitations and information of AS355 N helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

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6. Helicopter model AS 355NP

Type Certificate Holder:	Airbus Helicopters Aéroport International Marseille Provence 13725 Marignane, Cedex, France
Manufacturer:	Airbus Helicopters Aéroport International Marseille-Provence 13725 Marignane, Cedex, France
Aircraft description	Single-rotor helicopter with tail rotor, equipped with two gas-turbine engines and skid landing gear
Category	Normal
Applicability	AS 355NP helicopter model is approved for VFR and IFR day and night operation, for passenger transportation, for cargo transportation inside cabin as well as on external sling
Type Certificate Data	Type Certificate № 112-355 Issued by IAC AR on 06 December 1996
Type design	Defined in the document «355ABN0048 - AS355 FATA Type Design Definition», issue L
Certification basis	CE355.27 Certification Basis includes requirements to Airworthiness AP-27, requirements to Environment AP-36
Engine	Two gas-turbine engines Arrius 1A manufactured by Turbomeca Engine Type Certificate № 111-Д/01 dated on 28 October 1996 Issued by IAC AR
Noise requirements	Noise Supplement Type Certificate № CIII-102-AS355/Д01 dated 28 February 2008
Fuel	PT, TC-1 in accordance with GOST 10227-86 (foreign fuels types are listed in RFM) Anti-ice additives: fluid "И" (GOST 8313), "И-М" (TU 6-10-1458), volume concentration 0.10-0.30% %. Anti-static Sigbol is approved for use in amounts up to 0.0005% by weight.
Approved oil types for engine and transmission gearbox	See RFM

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Arrius 1A engine operational limits:

AOE:

Take-off (5 min)	
Power	934 h.p.(686 kW)
Generator speed	54375 rpm (100.5%)
Gas temperature before turbine	773 °C

Maximum continuous mode	
Power	830 h.p. (610 kW)
Generator speed	53397 rpm (98,7%)
Gas temperature before turbine	749 °C

100%=54177 rpm

OEI:

Maximum mode (2,5 minutes power)	
Power	564 h.p.(415 kW)
Generator speed	56347 rpm (104,1%)
Gas temperature before turbine	885 °C

Maximum continuous mode	
Power	525 h.p.(386 kW)
Generator speed	55452 rpm (102,5%)
Gas temperature before turbine	812 °C

Information on limitations on the transitional modes is contained in the approved RFM. Other limitations on the engine are given in the Type Certificate Data Sheet №1111-Д/01 dated on 28 October 1996 issued by IAC AR on engines TURBOMECA Arrius 1A.

Rotor Limitations

Power on flight	375 - 394 rpm
Maximum in autorotation	425 rpm
Minimum in autorotation	330 rpm

Speed limitations

Vne is limited by indicated air speed:
150 kt (287 km/h).
Vne at autorotation speed:
120 kt (222 km/h)

C.G. Reference

Longitudinal: - 3.4 m forward of the MRH centerline

Lateral: Aircraft symmetry plane

Minimum crew

1 pilot in R.H. seat

Title	Issue	Date
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Maximum take-off weight	2600 kg
Fuel capacity	736,7 liters
Number of seats	6 (including pilot seat)
Maximum baggage weight	In R.H. side hold 100 kg In L.H. side hold 120 kg In rear hold 80 kg On cabin floor: Forward section 150 kg Rear section 310 kg
Maximum operational altitude	6096 m
OAT temperature range	-40 °C...MCA +35 °C (Max +50 °C)

Additional operational conditions limitations and information of AS355 NP helicopter model for operators in Russian Federation:

1. Flights in icing conditions are prohibited.
2. Regular commercial transportation on helicopters, not equipped with a flight data recorder, is prohibited.
3. Flights in thunderstorm activity when weather radar is not installed or inoperative are prohibited.
4. Non-hangar storage helicopter operation is allowed only with use of protective covers and gags.

Other limitations are contained in helicopter operational documentations

Supplemental Type Certificates (STC)

No	STC Name	STC Holder	Type Design description documents	Aviation authorities issued STC	Applicability
1	STC SH 93-4 «Bearpaw Installation»	Dart Aerospace Ltd.	- MDL-D350-578, Rev.A; - Maintenance ICA-D350-578, Rev.1; - Installation Drawing D350-578, Rev. F	TCCA	AS355E AS355F AS355F1 AS355F2 AS355N AS355NP
2	STC SH 94-14 «Heli-utility Basket Installation»	Dart Aerospace Ltd.	- MDL-D350-607, Rev.A; - Maintenance ICA-D350-607, Rev.H; - Installation D350-578 Rev.H; - FMS-D355-607, Rev.D	TCCA	AS355E AS355F AS355F1 AS355F2 AS355N AS355NP

Title	Issue	Date
Data Sheet № FATA-AS355	02	30.03.2018

Supplements to Type Certificate and Major change approvals

Major change approvals	Type design change description	Applicability
№112-355/1	Introduction of the AS355E, F, F1, F2 models	AS355E AS355F AS355F1 AS355F2
№112-355/Д02	Installation of Turbomeca Arrius 1A1 engine	AS355NP
№112-355/Д03	Helicopter equipment Installation which provides IFR operation	AS355NP
№112-355/ОГЛ-04	New main gear box fixed ring	AS355NP
	Installation of new flight servo control NOVINTEC	
	Wire protection of strobe light	
	Double locking wire for yaw control	
№112-355/ОГИ-05	Long tube of tail drive line by air furnace	AS355NP
№112-355/ОГИ-06	Upper scissor branch by air furnace	AS355NP
№112-355/ОГИ-07	Engine fire extinction circuit inversion	AS355NP
№112-355/ОГИ-08	Free wheel pinion with full radius teeth	AS355NP
№FATA-02053R-MC-09	VEVD NG – Twin engines	AS355NP
№FATA-02053R-MC-10	Modification of the types of Chin weights of tail rotors	AS355N AS355NP
№FATA-02053R-MC-11	Half laminated bearings classification modification	AS355E AS355F AS355F1 AS355F2 AS355N AS355NP
№FATA-02053R-MC-12	Cargo compartment / Tail-boom junction frame	AS355E AS355F AS355F1 AS355F2 AS355N
№FATA-020165R-MC-13	Standardization of MGB main housing blanks	AS355E AS355F AS355F1 AS355F2 AS355N AS355NP

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Original TCDS is signed by Deputy Director General

Mr. O. Storchevov